TRMM Microwave Imager Microwave Brightness Temperature Legacy Product version-8

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TRMM Microwave Imager (TMI) Legacy Brightness Temperature Product version-8

- The entire > 17-year TMI time series has been reprocessed to produce the V8 Tb product
- To improve the quality of this legacy product, major revisions of the former 1B11 "Rad_counts-to-Tb" processing software were implemented
- This paper briefly summaries this significant XCAL accomplishment

History of TMI 1B11

- During on-orbit start-up, TMI measurements of the oceans were discovered to be too warm
 - A "deep space calibration" maneuver was performed and the TMI reflector was found to be slightly emissive (~3%); therefore

$$T_{ant} = (1 - \epsilon) * T_{earth_{scene}} + (\epsilon) * T_{phy}$$

History of TMI 1B11

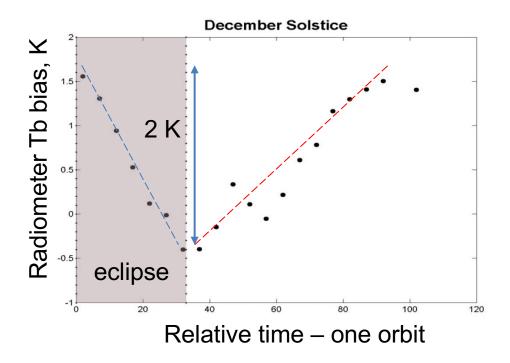
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- During the TRMM Cal/Val period, the <u>orbit mean bias</u> was removed by inter-calibration with SSMI
- This became 1B11 Version-6

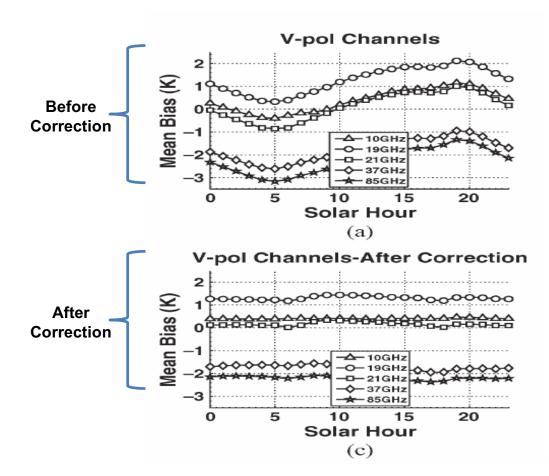
History of TMI 1B11 cont.

• Unfortunately, the reflector T_{phy} varied over the orbit and it was not measured, which produced a timevariable Tb error in 1B11 V6, which was ignored



History of TMI 1B11 cont.-2

- In 2004, XCAL was formed and this calibration error was rediscovered during inter-satellite comparisons
 - Based upon Channel 10V, an empirical correction was implemented in 1B11 V7



TMI Brightness Temps V8 Objectives

- To replace all previous ad hoc corrections with a new rigorous Rad_counts-to-Tb algorithm based upon on-orbit measurements and EM theory
- The major changes included:
 - Improved spacecraft attitude → TMI Geolocation
 - Cross-track (azimuth scan angle) Tb bias correction
 - TMI emissive reflector correction
 - TMI hot-load correction (new)
 - RFI in cold reflector correction (new)

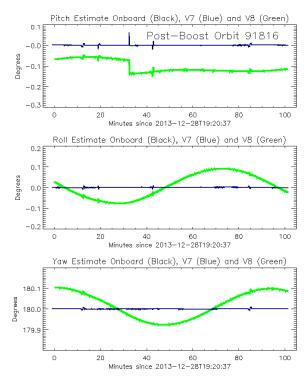
TRMM Spacecraft Attitude History Recalculated using PR roll estimates, Sun-Sensor and Gyro data

Sample Orbits

Preboost: errors up to 0.4° due to Sun effect on Earth Sensor

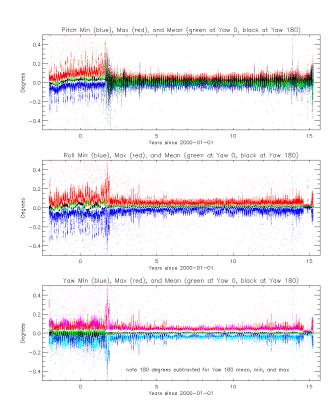
Pitch Estimate Onboard (Black), V7 (Blue) and V8 (Green) 0.4 Pre-boost Orbit 21251 0.2 Minutes since 2001-08-06T11:28:24.339 Roll Estimate Onboard (Black), V7 (Blue) and V8 (Green) 0.3 0.2 -0.260 Minutes since ATT file start Yaw Estimate Onboard (Black), V7 (Blue) and V8 (Green) 0.3 0.2 -0.2-0.3Minutes since 2001-08-06T11:28:24.339

Postboost: errors often > 0.1° with pitch drift and sinusoidal roll/yaw



Trends in Min/Max/Mean

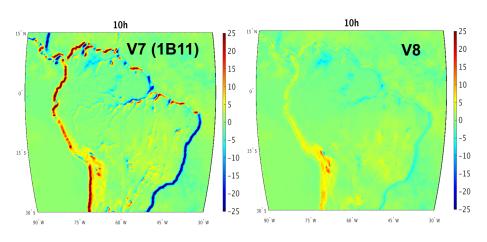
Motion now tracked to < 0.01° showing true attitude variation



TMI Geolocation and Cross-Track Bias Corrections

TMI Pointing Accuracy

- Spin axis alignment (pitch/roll offsets) derived from analysis of cross-track bias patterns
- Feed horn mount offsets (cone angle and azimuth start angle) derived from coastline analysis of yaw-0 versus yaw-180 differences

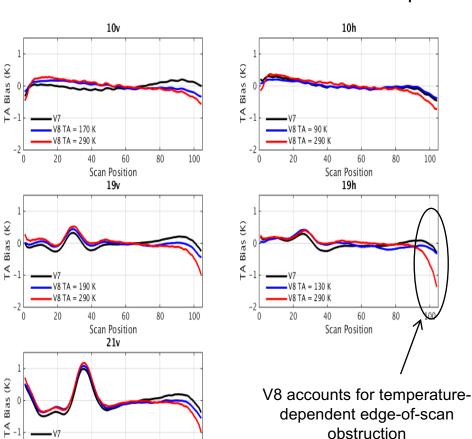


Delta Cone & Azimuth Start Angle Offsets from V7 to V8

Channel	Cone Angle		Cone Angle Offset for EIA	Azimuth Start Angle	
	V7	V8	V8	V7	V8
10 GHz	49.0	49.45	10v: -0.048 10h: +0.048	-64.4024	-63.91
19, 21, 37, 85 GHz	49.0	49.28	0	-64.4024	-64.36

Cross-Track Bias Correction

•V8 derived along-scan correction as a function of temperature, using a linear interpolation between vicarious cold and warm techniques

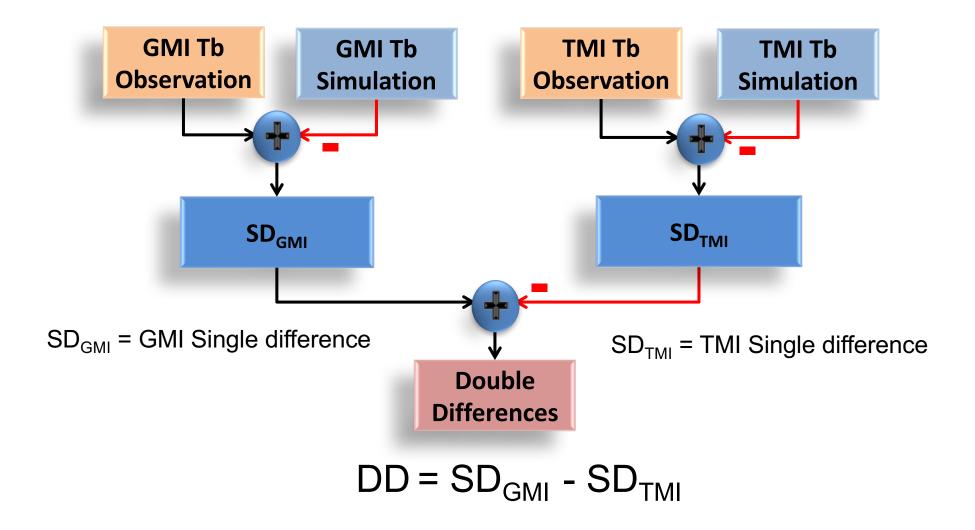


V8 TA = 210 K

Scan Position

XCAL Radiometric Double Differences

GPM provided 13 month overlap with TRMM



Emissive Reflector Correction Procedure

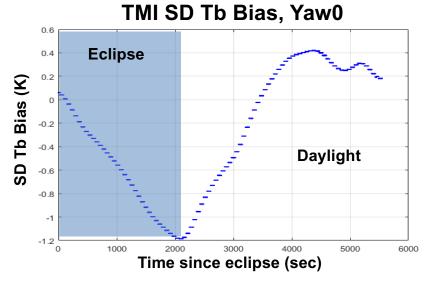
- Measure the reflector emissivity for each TMI channel
 - A special deep space calibration maneuver @ end of TRMM mission
- Calculate Earth scene brightness

$$T_{earth_scene} = \frac{T_{ant} - \epsilon T_{phy}}{(1 - \epsilon)}$$
Not measured

TMI Reflector Physical Temperature Derivation

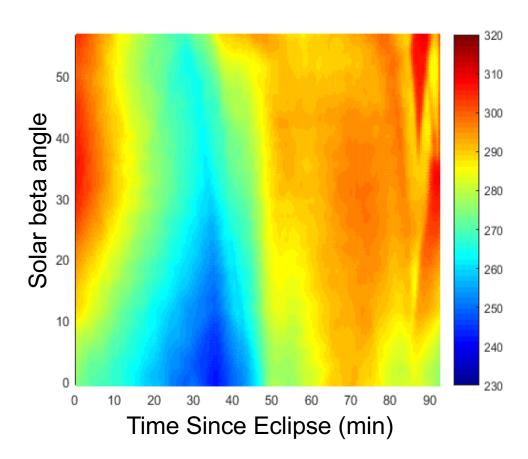
•Reflector T_{phy} derived during normal science operations using 10V channel brightness SD_{TMI}

$$T_{phy} = const * SD_{TIM} / (emissivity_{10V})$$

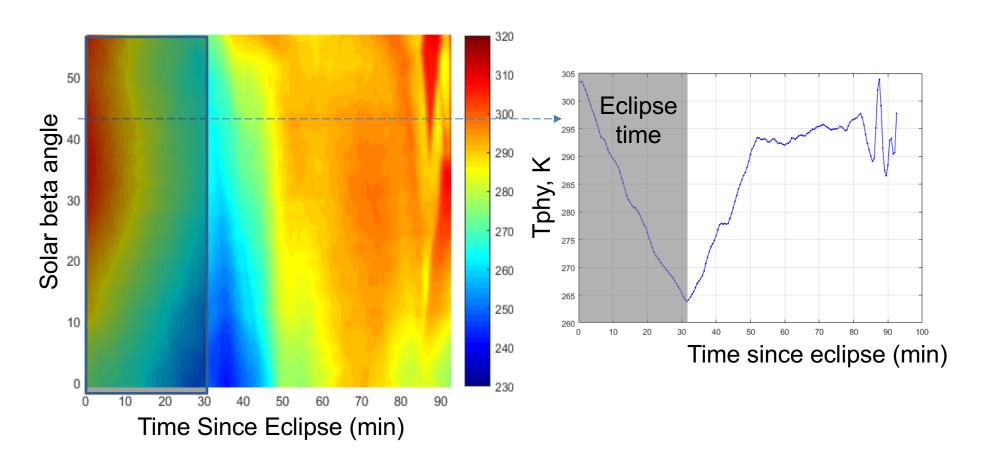


- Data collected for 10-years
 - Separately for yaw0 & yaw180
 - Gridded as a function of solar beta angle (0.25 deg) and time since eclipse (30 sec)

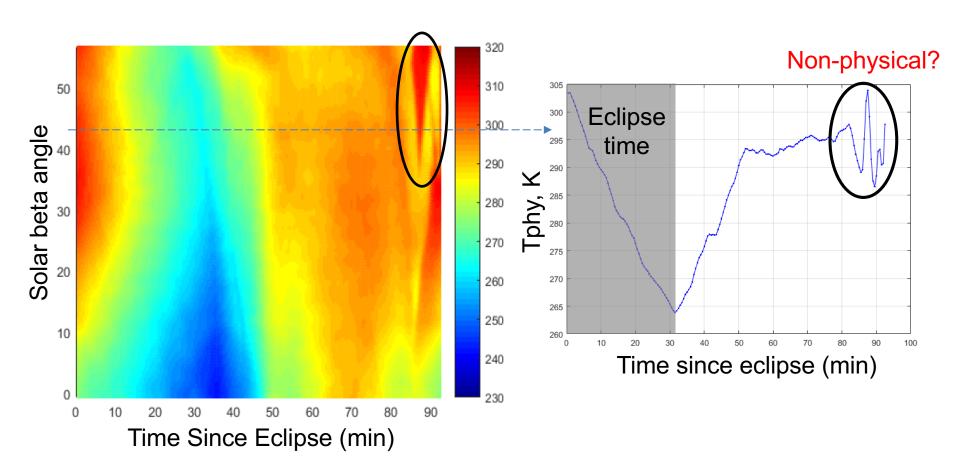
 T_{phy} was derived as a function of solar beta angle and time since entering eclipse



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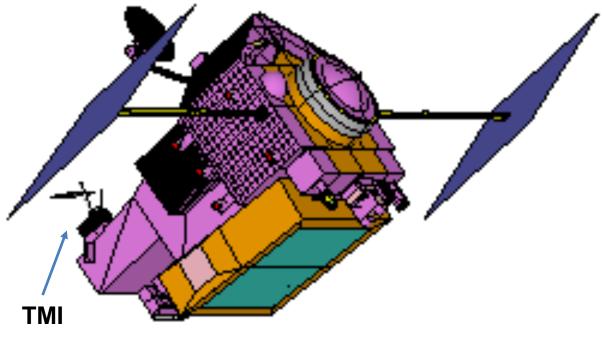


 T_{phy} was derived as a function of solar beta angle and time since entering eclipse



Geometry of Solar Intrusion on TMI Hot Load





TMI Instrument showing hot load at center.

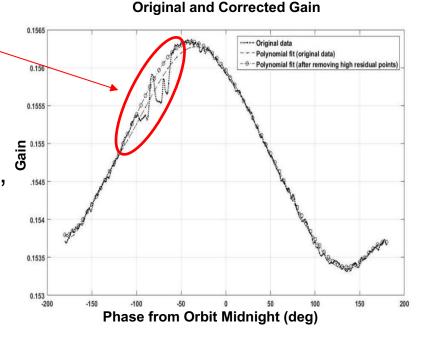
Spacecraft as seen from Sun at high solar beta angle before Earth shadow entry.

TMI Hot-Load Correction

At high solar beta angle, solar intrusions onto TMI hot-load occurred, which resulted in an anomalous

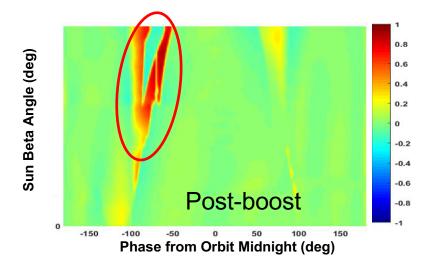
gain calculation

To remove this error, we introduced correction of the hot load temp (ΔT_h) , which was added to the hot load physical temp measurement

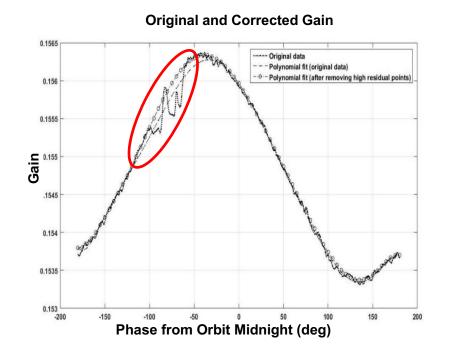


TMI Hot-Load Correction

Correction for solar intrusions into TMI hot-load implemented for V8 were applied as function of the Solar Beta Angle and Phase from Orbit Midnight (related to time since eclipse)

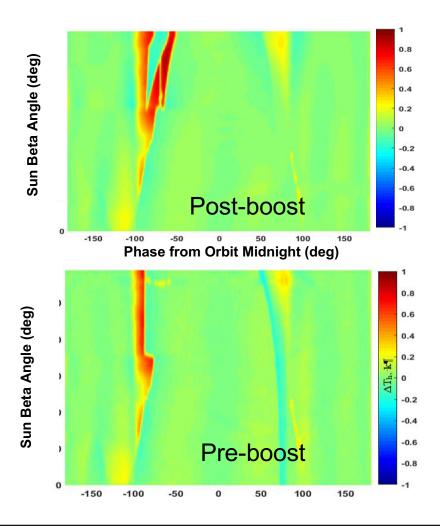


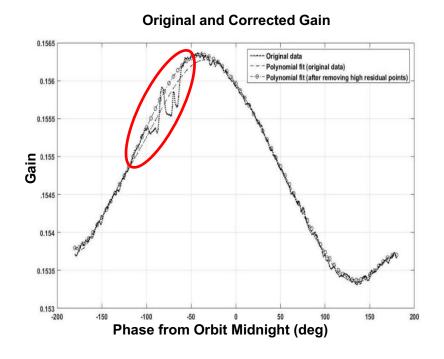
hot load temperature Correction (ΔT_h)



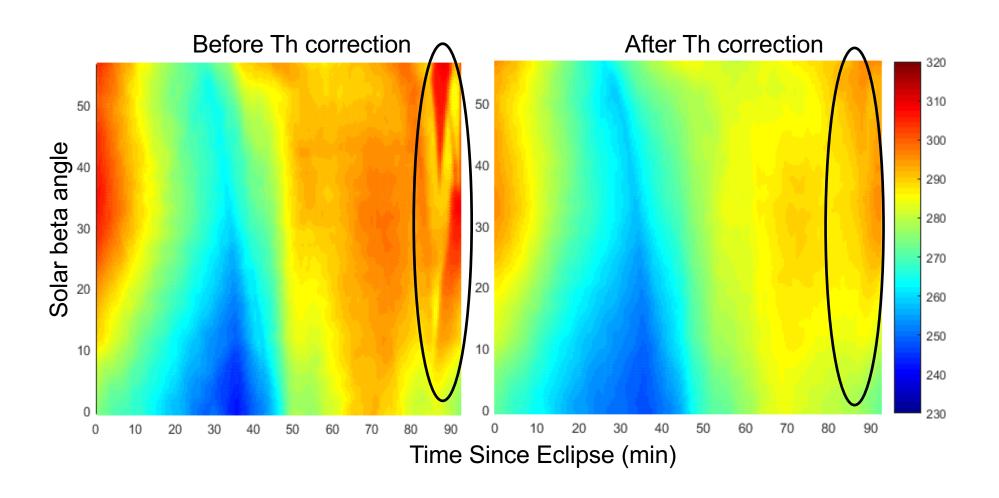
TMI Hot-Load Correction

Correction for solar intrusions into TMI hot-load implemented for V8 were different for pre- and post-boost



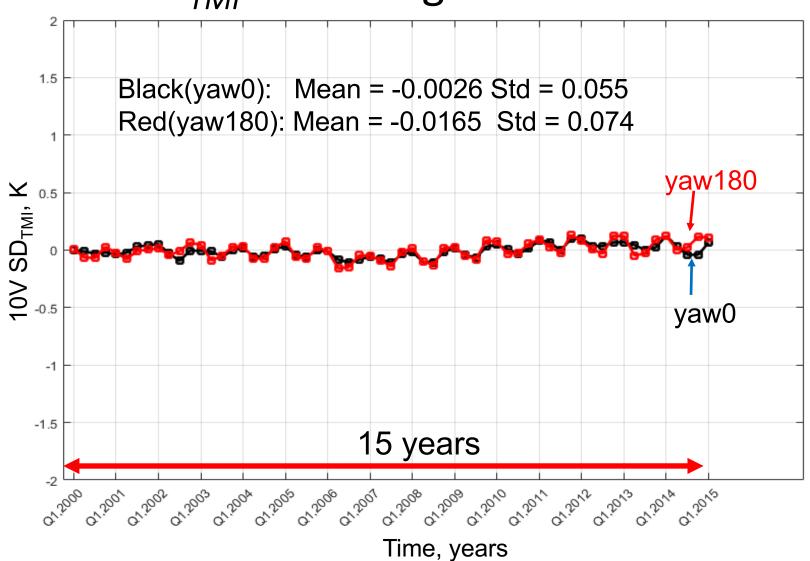


After applying the hot load correction, Tphy was rederived



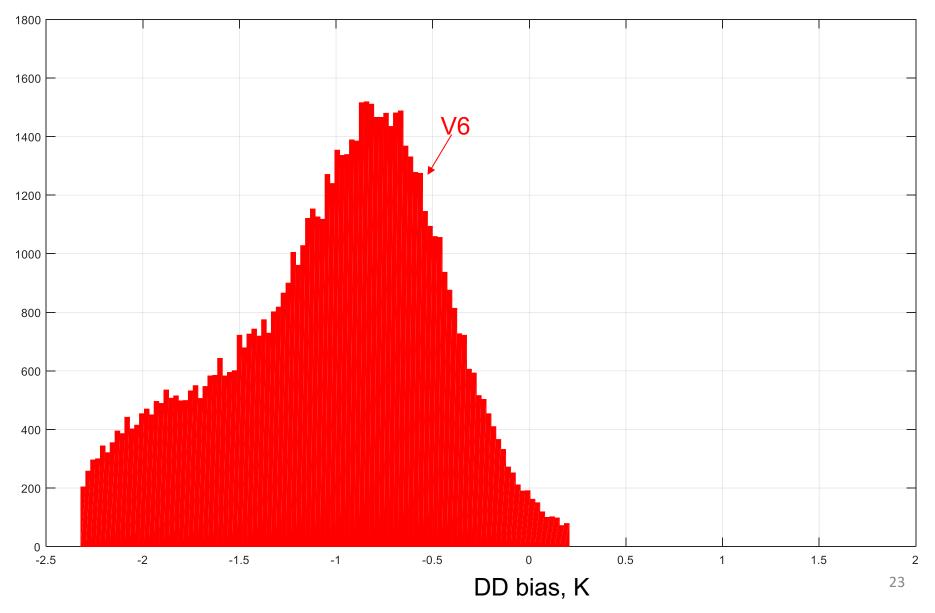
TMI Tb V8 Validation

TMI V8 Tb Radiometric Stability $10V SD_{TMI}$: 3-mo avg for 2000 - 2015



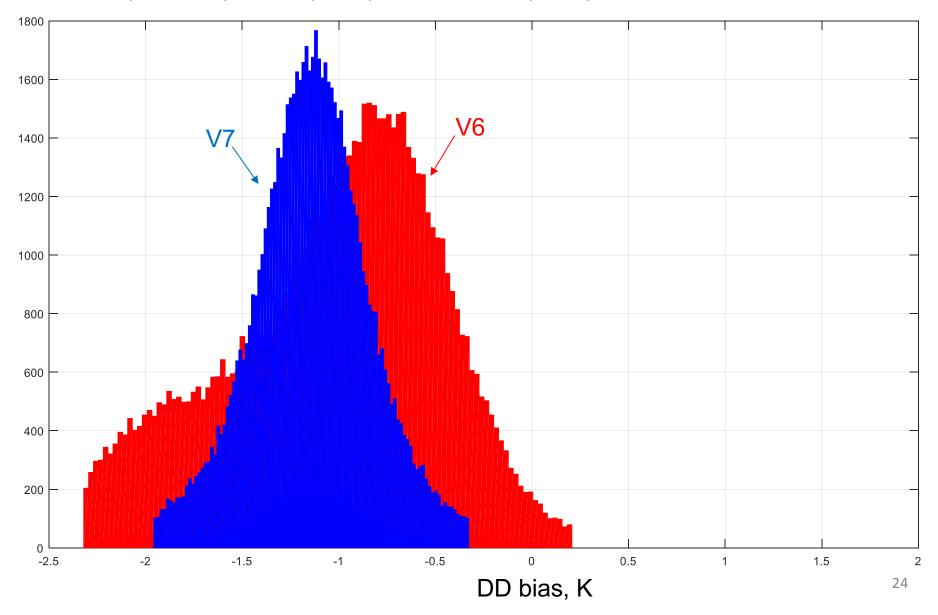
10V Chan DD Histograms (TMI & GMI)

GMI (ITE101) , TMI(V06): Red,



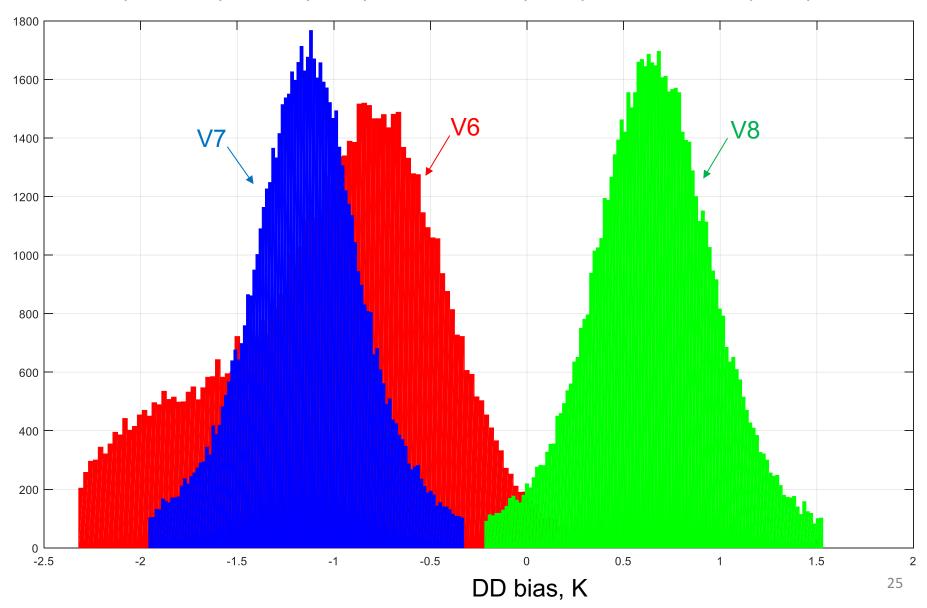
10V Chan DD Histograms (TMI & GMI)

• GMI (ITE101), TMI(V06): Red, TMI(V07): Blue,



10V Chan DD Histograms (TMI & GMI)

• GMI (ITE101), TMI(V06): Red, TMI(V07): Blue, TMI(V08): Green



Conclusions

- Development of legacy TMI version-8 Tb product is an extraordinary accomplishment of the XCAL team
- 1,000's hrs of collaborative effort have resulted in a TMI climate data record time, which will be used to calibrate the TRMM era constellation radiometers
- Provides a seamless transition of GPROF precipitation measurements from TRMM to GPM